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Reply to Office Action of April 29, 2005  
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**Amendments to the Specification:**

Please replace the second paragraph of the Detailed Description on page 3, lines 22-31, with the following amended paragraph:

To assemble the aircraft-side part 12, as shown in Figure 2, the attachment ring 20 is slid over the barrel up to a shoulder 38 on the core 14. The attachment ring 20 is shown with a lanyard 22 fitted. However, alternate embodiments could include an alternative cable strain relief device or interface. The next component fitted over the barrel 16 is mounting ring 26 and fingers 24, which are pre-assembled on the sleeve 28 together with an electro-magnetic compatibility (EMC) shielding ring 32. This assembly 24, 26, 28 is retained over the barrel by the clamping ring 30, which is screwed onto external threads on the core 14. Tightening the ring 30 clamps the sleeve 28 and mounting ring ~~28~~ 26 against a further shoulder 40 formed on the core 14. The attachment ring 20 is thereby trapped for free rotation on the core 14 between the shoulder 38 and the mounting ring 26.